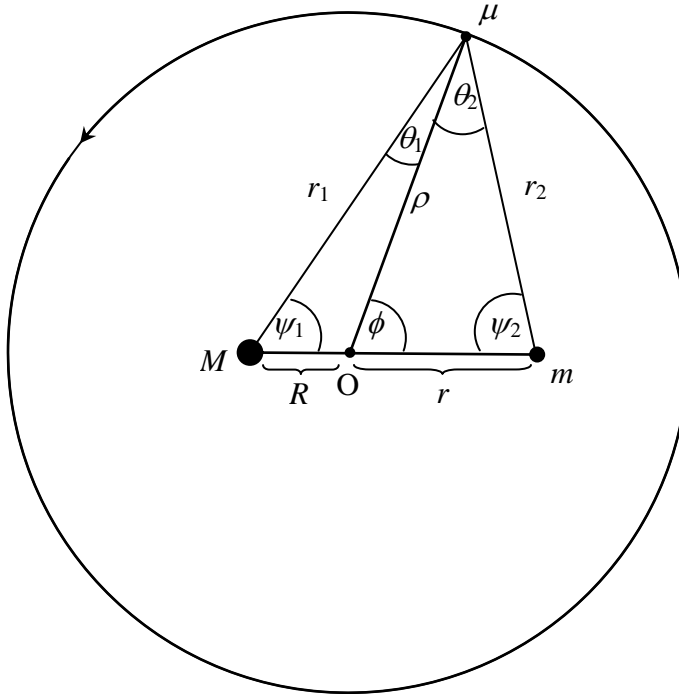


I. Solution



1.1 Let O be their centre of mass. Hence

$$MR - mr = 0 \quad \dots\dots\dots (1)$$

$$m\omega_0^2 r = \frac{GMm}{(R+r)^2} \quad \dots\dots\dots (2)$$

$$M\omega_0^2 R = \frac{GMm}{(R+r)^2}$$

From Eq. (2), or using reduced mass, $\omega_0^2 = \frac{G(M+m)}{(R+r)^3}$

$$\text{Hence, } \omega_0^2 = \frac{G(M+m)}{(R+r)^3} = \frac{GM}{r(R+r)^2} = \frac{Gm}{R(R+r)^2} \quad \dots\dots\dots (3)$$

1.2 Since μ is infinitesimal, it has no gravitational influences on the motion of neither M nor m . For μ to remain stationary relative to both M and m we must have:

$$\frac{GM\mu}{r_1^2} \cos \theta_1 + \frac{Gm\mu}{r_2^2} \cos \theta_2 = \mu \omega_0^2 \rho = \frac{G(M+m)\mu}{(R+r)^3} \rho \quad \dots\dots\dots (4)$$

$$\frac{GM\mu}{r_1^2} \sin \theta_1 = \frac{Gm\mu}{r_2^2} \sin \theta_2 \quad \dots\dots\dots (5)$$

Substituting $\frac{GM}{r_1^2}$ from Eq. (5) into Eq. (4), and using the identity

$\sin \theta_1 \cos \theta_2 + \cos \theta_1 \sin \theta_2 = \sin(\theta_1 + \theta_2)$, we get

$$m \frac{\sin(\theta_1 + \theta_2)}{r_2^2} = \frac{(M+m)}{(R+r)^3} \rho \sin \theta_1 \quad \dots\dots\dots (6)$$

The distances r_2 and ρ , the angles θ_1 and θ_2 are related by two Sine Rule equations

$$\frac{\sin \psi_1}{\rho} = \frac{\sin \theta_1}{R} \quad \dots\dots\dots (7)$$

$$\frac{\sin \psi_1}{r_2} = \frac{\sin(\theta_1 + \theta_2)}{R+r}$$

Substitute (7) into (6)

$$\frac{1}{r_2^3} = \frac{R}{(R+r)^4} \frac{(M+m)}{m} \quad \dots\dots\dots (10)$$

Since $\frac{m}{M+m} = \frac{R}{R+r}$, Eq. (10) gives

$$r_2 = R+r \quad \dots\dots\dots (11)$$

By substituting $\frac{Gm}{r_2^2}$ from Eq. (5) into Eq. (4), and repeat a similar procedure, we get

$$r_1 = R+r \quad \dots\dots\dots (12)$$

Alternatively,

$$\frac{r_1}{\sin(180^\circ - \phi)} = \frac{R}{\sin \theta_1} \quad \text{and} \quad \frac{r_2}{\sin \phi} = \frac{r}{\sin \theta_2}$$

$$\frac{\sin \theta_1}{\sin \theta_2} = \frac{R}{r} \times \frac{r_2}{r_1} = \frac{m}{M} \times \frac{r_2}{r_1}$$

Combining with Eq. (5) gives $r_1 = r_2$

Hence, it is an equilateral triangle with

$$\begin{aligned} \psi_1 &= 60^\circ \\ \psi_2 &= 60^\circ \end{aligned} \dots\dots\dots (13)$$

The distance ρ is calculated from the Cosine Rule.

$$\begin{aligned} \rho^2 &= r^2 + (R+r)^2 - 2r(R+r) \cos 60^\circ \\ \rho &= \sqrt{r^2 + rR + R^2} \end{aligned} \dots\dots\dots (14)$$

Alternative Solution to 1.2

Since μ is infinitesimal, it has no gravitational influences on the motion of neither M nor m . For μ to remain stationary relative to both M and m we must have:

$$\frac{GM\mu}{r_1^2} \cos \theta_1 + \frac{Gm\mu}{r_2^2} \cos \theta_2 = \mu \omega^2 \rho = \frac{G(M+m)\mu}{(R+r)^3} \rho \dots\dots\dots (4)$$

$$\frac{GM\mu}{r_1^2} \sin \theta_1 = \frac{Gm\mu}{r_2^2} \sin \theta_2 \dots\dots\dots (5)$$

Note that

$$\begin{aligned} \frac{r_1}{\sin(180^\circ - \phi)} &= \frac{R}{\sin \theta_1} \\ \frac{r_2}{\sin \phi} &= \frac{r}{\sin \theta_2} \quad (\text{see figure}) \\ \frac{\sin \theta_1}{\sin \theta_2} &= \frac{R}{r} \times \frac{r_2}{r_1} = \frac{m}{M} \times \frac{r_2}{r_1} \end{aligned} \dots\dots\dots (6)$$

Equations (5) and (6): $r_1 = r_2 \dots\dots\dots (7)$

$$\frac{\sin \theta_1}{\sin \theta_2} = \frac{m}{M} \dots\dots\dots (8)$$

$$\psi_1 = \psi_2 \dots\dots\dots (9)$$

The equation (4) then becomes:

$$M \cos \theta_1 + m \cos \theta_2 = \frac{(M+m)}{(R+r)^3} r_1^2 \rho \dots\dots\dots (10)$$

Equations (8) and (10): $\sin(\theta_1 + \theta_2) = \frac{M+m}{M} \frac{r_1^2 \rho}{(R+r)^3} \sin \theta_2 \dots\dots\dots (11)$

Note that from figure, $\frac{\rho}{\sin \psi_2} = \frac{r}{\sin \theta_2} \dots\dots\dots (12)$

Equations (11) and (12): $\sin(\theta_1 + \theta_2) = \frac{M+m}{M} \frac{r_1^2 r}{(R+r)^3} \sin \psi_2$ (13)

Also from figure,

$$(R+r)^2 = r_2^2 - 2r_1 r_2 \cos(\theta_1 + \theta_2) + r_1^2 = 2r_1^2 [1 - \cos(\theta_1 + \theta_2)]$$
 (14)

Equations (13) and (14): $\sin(\theta_1 + \theta_2) = \frac{\sin \psi_2}{2[1 - \cos(\theta_1 + \theta_2)]}$ (15)

$$\theta_1 + \theta_2 = 180^\circ - \psi_1 - \psi_2 = 180^\circ - 2\psi_2 \quad (\text{see figure})$$

$$\therefore \cos \psi_2 = \frac{1}{2}, \psi_2 = 60^\circ, \psi_1 = 60^\circ$$

Hence M and m form an equilateral triangle of sides $(R+r)$

Distance μ to M is $R+r$

Distance μ to m is $R+r$

Distance μ to O is $\rho = \sqrt{\left(\frac{R+r}{2} - R\right)^2 + \left\{(R+r)\frac{\sqrt{3}}{2}\right\}^2} = \sqrt{R^2 + Rr + r^2}$

1.3 The energy of the mass μ is given by

$$E = -\frac{GM\mu}{r_1} - \frac{Gm\mu}{r_2} + \frac{1}{2}\mu\left(\left(\frac{d\rho}{dt}\right)^2 + \rho^2\omega^2\right)$$
(15)

Since the perturbation is in the radial direction, angular momentum is conserved

($r_1 = r_2 = \mathfrak{R}$ and $m = M$),

$$E = -\frac{2GM\mu}{\mathfrak{R}} + \frac{1}{2}\mu\left(\left(\frac{d\rho}{dt}\right)^2 + \frac{\rho_0^4 \omega_0^2}{\rho^2}\right)$$
(16)

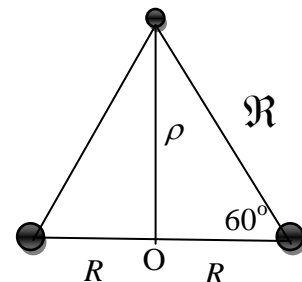
Since the energy is conserved,

$$\frac{dE}{dt} = 0$$

$$\frac{dE}{dt} = \frac{2GM\mu}{\mathfrak{R}^2} \frac{d\mathfrak{R}}{dt} + \mu \frac{d\rho}{dt} \frac{d^2\rho}{dt^2} - \mu \frac{\rho_0^4 \omega_0^2}{\rho^3} \frac{d\rho}{dt} = 0$$
(17)

$$\frac{d\mathfrak{R}}{dt} = \frac{d\mathfrak{R}}{d\rho} \frac{d\rho}{dt} = \frac{d\rho}{dt} \frac{\rho}{\mathfrak{R}}$$
(18)

$$\frac{dE}{dt} = \frac{2GM\mu}{\mathfrak{R}^3} \rho \frac{d\rho}{dt} + \mu \frac{d\rho}{dt} \frac{d^2\rho}{dt^2} - \mu \frac{\rho_0^4 \omega_0^2}{\rho^3} \frac{d\rho}{dt} = 0$$
(19)



Since $\frac{d\rho}{dt} \neq 0$, we have

$$\frac{2GM}{\mathfrak{R}^3} \rho + \frac{d^2\rho}{dt^2} - \frac{\rho_0^4 \omega_0^2}{\rho^3} = 0 \text{ or}$$

$$\frac{d^2\rho}{dt^2} = -\frac{2GM}{\mathfrak{R}^3} \rho + \frac{\rho_0^4 \omega_0^2}{\rho^3}. \quad \dots\dots\dots(20)$$

The perturbation from \mathfrak{R}_0 and ρ_0 gives $\mathfrak{R} = \mathfrak{R}_0 \left(1 + \frac{\Delta\mathfrak{R}}{\mathfrak{R}_0}\right)$ and $\rho = \rho_0 \left(1 + \frac{\Delta\rho}{\rho_0}\right)$.

Then

$$\frac{d^2\rho}{dt^2} = \frac{d^2}{dt^2}(\rho_0 + \Delta\rho) = -\frac{2GM}{\mathfrak{R}_0^3 \left(1 + \frac{\Delta\mathfrak{R}}{\mathfrak{R}_0}\right)^3} \rho_0 \left(1 + \frac{\Delta\rho}{\rho_0}\right) + \frac{\rho_0^4 \omega_0^2}{\rho_0^3 \left(1 + \frac{\Delta\rho}{\rho_0}\right)^3} \quad \dots\dots\dots(21)$$

Using binomial expansion $(1 + \varepsilon)^n \approx 1 + n\varepsilon$,

$$\frac{d^2\Delta\rho}{dt^2} = -\frac{2GM}{\mathfrak{R}_0^3} \rho_0 \left(1 + \frac{\Delta\rho}{\rho_0}\right) \left(1 - \frac{3\Delta\mathfrak{R}}{\mathfrak{R}_0}\right) + \rho_0 \omega_0^2 \left(1 - \frac{3\Delta\rho}{\rho_0}\right). \quad \dots\dots\dots(22)$$

Using $\Delta\rho = \frac{\mathfrak{R}}{\rho} \Delta\mathfrak{R}$,

$$\frac{d^2\Delta\rho}{dt^2} = -\frac{2GM}{\mathfrak{R}_0^3} \rho_0 \left(1 + \frac{\Delta\rho}{\rho_0} - \frac{3\rho_0 \Delta\rho}{\mathfrak{R}_0^2}\right) + \rho_0 \omega_0^2 \left(1 - \frac{3\Delta\rho}{\rho_0}\right). \quad \dots\dots\dots(23)$$

Since $\omega_0^2 = \frac{2GM}{\mathfrak{R}_0^3}$,

$$\frac{d^2\Delta\rho}{dt^2} = -\omega_0^2 \rho_0 \left(1 + \frac{\Delta\rho}{\rho_0} - \frac{3\rho_0 \Delta\rho}{\mathfrak{R}_0^2}\right) + \omega_0^2 \rho_0 \left(1 - \frac{3\Delta\rho}{\rho_0}\right) \quad \dots\dots\dots(24)$$

$$\frac{d^2\Delta\rho}{dt^2} = -\omega_0^2 \rho_0 \left(\frac{4\Delta\rho}{\rho_0} - \frac{3\rho_0 \Delta\rho}{\mathfrak{R}_0^2}\right) \quad \dots\dots\dots(25)$$

$$\frac{d^2\Delta\rho}{dt^2} = -\omega_0^2 \Delta\rho \left(4 - \frac{3\rho_0^2}{\mathfrak{R}_0^2}\right) \quad \dots\dots\dots(26)$$

From the figure, $\rho_0 = \mathfrak{R}_0 \cos 30^\circ$ or $\frac{\rho_0^2}{\mathfrak{R}_0^2} = \frac{3}{4}$,

$$\frac{d^2\Delta\rho}{dt^2} = -\omega_0^2 \Delta\rho \left(4 - \frac{9}{4}\right) = -\frac{7}{4} \omega_0^2 \Delta\rho. \quad \dots\dots\dots(27)$$

Angular frequency of oscillation is $\frac{\sqrt{7}}{2} \omega_0$.

Alternative solution:

$M = m$ gives $R = r$ and $\omega_0^2 = \frac{G(M+M)}{(R+R)^3} = \frac{GM}{4R^3}$. The unperturbed radial distance of μ is

$\sqrt{3}R$, so the perturbed radial distance can be represented by $\sqrt{3}R + \zeta$ where $\zeta \ll \sqrt{3}R$ as shown in the following figure.

Using Newton's 2nd law, $-\frac{2GM\mu}{\{R^2 + (\sqrt{3}R + \zeta)^2\}^{3/2}}(\sqrt{3}R + \zeta) = \mu \frac{d^2}{dt^2}(\sqrt{3}R + \zeta) - \mu\omega^2(\sqrt{3}R + \zeta)$.

(1)

The conservation of angular momentum gives $\mu\omega_0(\sqrt{3}R)^2 = \mu\omega(\sqrt{3}R + \zeta)^2$.

(2)

Manipulate (1) and (2) algebraically, applying $\zeta^2 \approx 0$ and binomial approximation.

$$-\frac{2GM}{\{R^2 + (\sqrt{3}R + \zeta)^2\}^{3/2}}(\sqrt{3}R + \zeta) = \frac{d^2\zeta}{dt^2} - \frac{\omega_0^2\sqrt{3}R}{(1 + \zeta/\sqrt{3}R)^3}$$

$$-\frac{2GM}{\{4R^2 + 2\sqrt{3}\zeta R\}^{3/2}}(\sqrt{3}R + \zeta) \approx \frac{d^2\zeta}{dt^2} - \frac{\omega_0^2\sqrt{3}R}{(1 + \zeta/\sqrt{3}R)^3}$$

$$-\frac{GM}{4R^3}\sqrt{3}R \frac{(1 + \zeta/\sqrt{3}R)}{(1 + \sqrt{3}\zeta/2R)^{3/2}} = \frac{d^2\zeta}{dt^2} - \frac{\omega_0^2\sqrt{3}R}{(1 + \zeta/\sqrt{3}R)^3}$$

$$-\omega_0^2\sqrt{3}R \left(1 - \frac{3\sqrt{3}\zeta}{4R}\right) \left(1 + \frac{\zeta}{\sqrt{3}R}\right) \approx \frac{d^2\zeta}{dt^2} - \omega_0^2\sqrt{3}R \left(1 - \frac{3\zeta}{\sqrt{3}R}\right)$$

$$\frac{d^2}{dt^2}\zeta = -\left(\frac{7}{4}\omega_0^2\right)\zeta$$

1.4 Relative velocity

Let v = speed of each spacecraft as it moves in circle around the centre O.

The relative velocities are denoted by the subscripts A, B and C.

For example, v_{BA} is the velocity of B as observed by A.

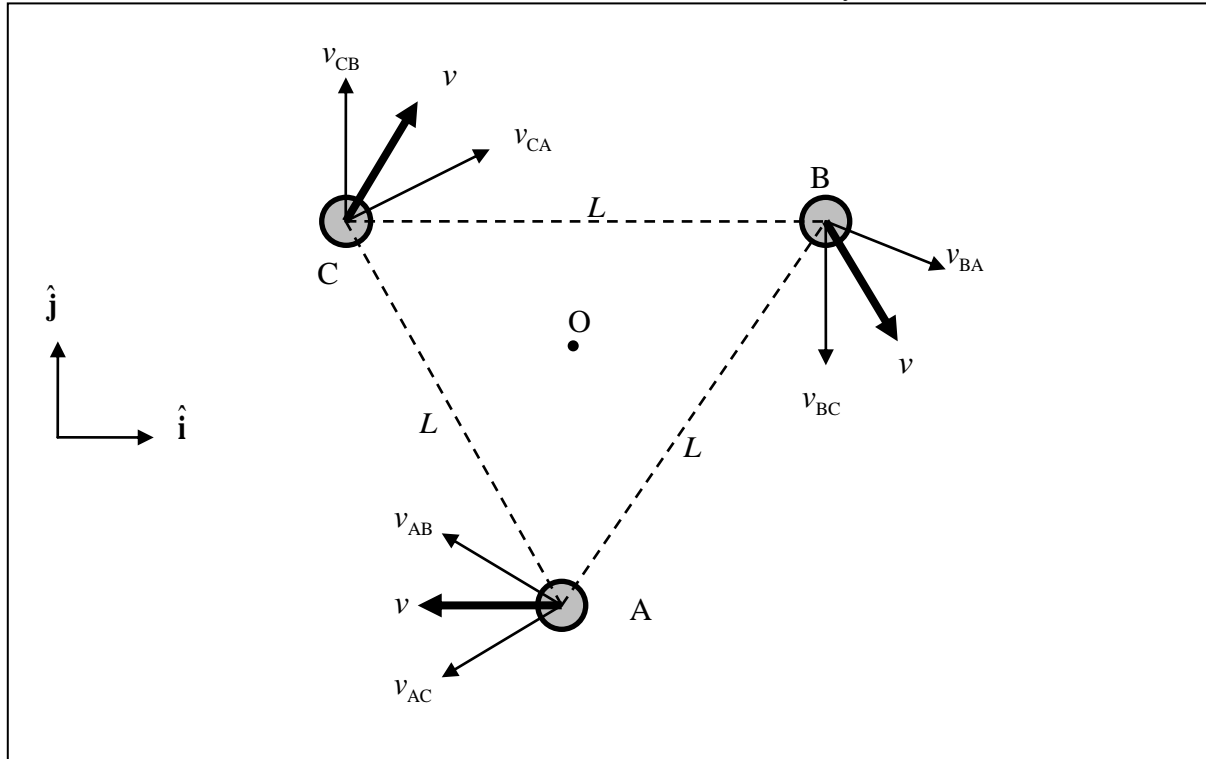
The period of circular motion is 1 year $T = 365 \times 24 \times 60 \times 60$ s. (28)

The angular frequency $\omega = \frac{2\pi}{T}$

The speed $v = \omega \frac{L}{2\cos 30^\circ} = 575$ m/s (29)

The speed is much less than the speed light \rightarrow Galilean transformation.

In Cartesian coordinates, the velocities of B and C (as observed by O) are



For B, $\vec{v}_B = v \cos 60^\circ \hat{i} - v \sin 60^\circ \hat{j}$

For C, $\vec{v}_C = v \cos 60^\circ \hat{i} + v \sin 60^\circ \hat{j}$

Hence $\vec{v}_{BC} = -2v \sin 60^\circ \hat{j} = -\sqrt{3}v \hat{j}$

The speed of B as observed by C is $\sqrt{3}v \approx 996 \text{ m/s}$ (30)

Notice that the relative velocities for each pair are anti-parallel.

Alternative solution for 1.4

One can obtain v_{BC} by considering the rotation about the axis at one of the spacecrafts.

$$v_{BC} = \omega L = \frac{2\pi}{365 \times 24 \times 60 \times 60 \text{ s}} (5 \times 10^6 \text{ km}) \approx 996 \text{ m/s}$$